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INTRODUCTION

- **Boeing** is interested in **additive manufacturing** with laser powder bed fusion (LPBF) for aerospace applications
- LPBF works by selectively fusing metal powder to form solid layers; powder then swept over to form more layers for a 3D part
- EOS M290 LPBF system under study • <u>Advs. of LPBF</u>: lower density & lighter weight, less wasted material, complex geometry capability
- <u>Current reliability issues in LPBF</u>: **local part porosity, material defects,** mechanical properties
- Argon gas (shielding gas to prevent oxidation) is used in manuf. process and is believed to be a main factor in reliability
- Project goal is to analyze and correlate Ar gas flow (process pressure setting) changes on test coupon material quality with gas dynamics in the chamber
- Seeking process pressure setting that produce the **most reliable quality** of LPBF metal parts to use in prototyping, low-vol. prod.

PROBLEM STATEMENT How can changing process pressure settings & gas flow improve LPBF metal quality (porosity)?

CORE FUNCTIONS & APPROACH



1:1 scale mockup chamber fabrication paralleled with design of experiment (DOE) based on design & logistical constraints, Boeing computational fluid dynamics (CFD), literature research, UW Round Robin data, and collaboration with Boeing & LPBF Interruption capstone builds



Powder Bed Fusion: Part Porosity Sources



MOCKUP DESIGN AND DEVELOPMENT

Mockup Chamber



Design & Manufacturing Steps

- Designed a CAD model of the chamber
- Applied DFMA to improve design concepts for producibility, ease of assembly
- and required tools and machines
- Manufactured components of the chamber at machine shop using waterjet, CNC mill, 3D printer, vertical bandsaw, drill press, etc
- Assembled the chamber, conducted leak test to ensure airtight and resolve technical issues

Gas Flow Validation







Boeing CFD model from right side (left); Zone 0 gas flow analysis (right)

Gas Flow Results

- Mockup validation setup a viable method of visualizing M290 gas flow
- of the Boeing CFD model
- Slight discrepancies in size of dead zones located in the center and bottom of the vertical planes
- Unable to quantitatively correlate the models due to equipment issues

• Developed manufacturing process: timeline, bills of materials, drawing files

Boeing CFD top-down model (top); lower horizontal plane gas flow analysis (bottom)

• All three vertical planes and horizontal planes largely matched that







- microscopy due to equipment failure

Coupon Analysis Results • Differences between zones or pressures in mechanical properties, fracture surface appearance, and porosity were **statistically**

- insignificant.
- consistently across metrics.

CONCLUSION & FUTURE WORK

- mechanical properties

Next Steps: Particle image velocimetry (PIV), additional builds at higher and lower pressure, micro-CT porosity analysis

Acknowledgements Boeing: Troy Haworth, Kevin Mejia, David Eckols, Stacey Huang, Cory Cunningham, Patrick Buffington UW: Dr. Dwayne Arola, Reid Schur, Dr. Owen Williams, Dr. Luna Huang, Dr. Eli Patten, Amy Lim, Bill Kyukendal, Hansen Fong, Jordan Hatch, UW Interruption Capstone Team

Seattle



COUPON RESULTS/VALIDATION

• **0.8 mbar** differential pressure test build **& 0.6 mbar** control

• Tensile results met strength specification but had varying ductility, and two outliers failed in grip section rather than gauge section (see fracture surfaces)



• MicroCT porosity analysis plan was designed and then **adapted to**

• ImageJ cross-section analysis provided quantitative porosity results • Fractures surfaces showed another perspective + **causes of outlier failures**

• Instead, variations within and between samples appeared

 CFD model validated, completing product of multi-year Boeing effort • Porosity, fracture and tensile properties analyzed for 0.6 and 0.8 mbar • Experiment can identify relationships between pressure, flow, and

Mechanical Engineering Capstone Exposition

June 2nd 2022, Husky Union Building, University of Washington,